

PROBABILISTIC DAMAGE TOLERANCE WITH BAYESIAN UPDATING

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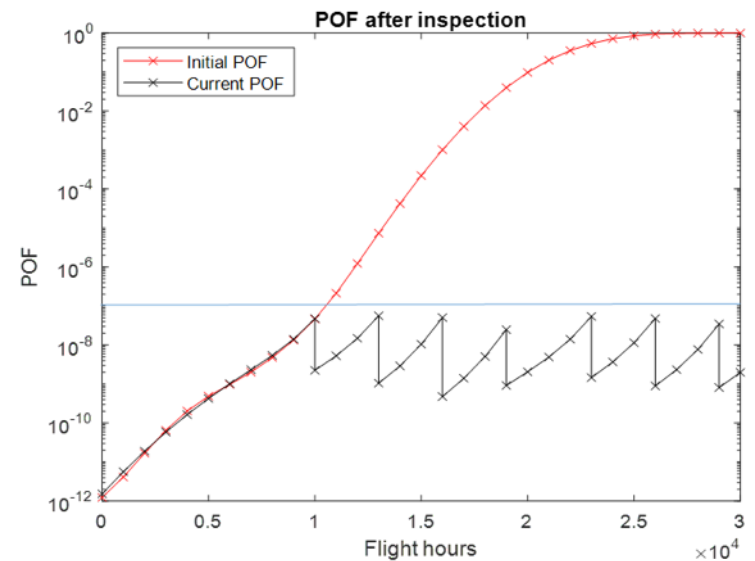




Overview



- Motivation
- SMART Probabilistic Damage Tolerance Analysis Quick Review
- Bayes Theorem
- Script
- Examples
- Conclusion



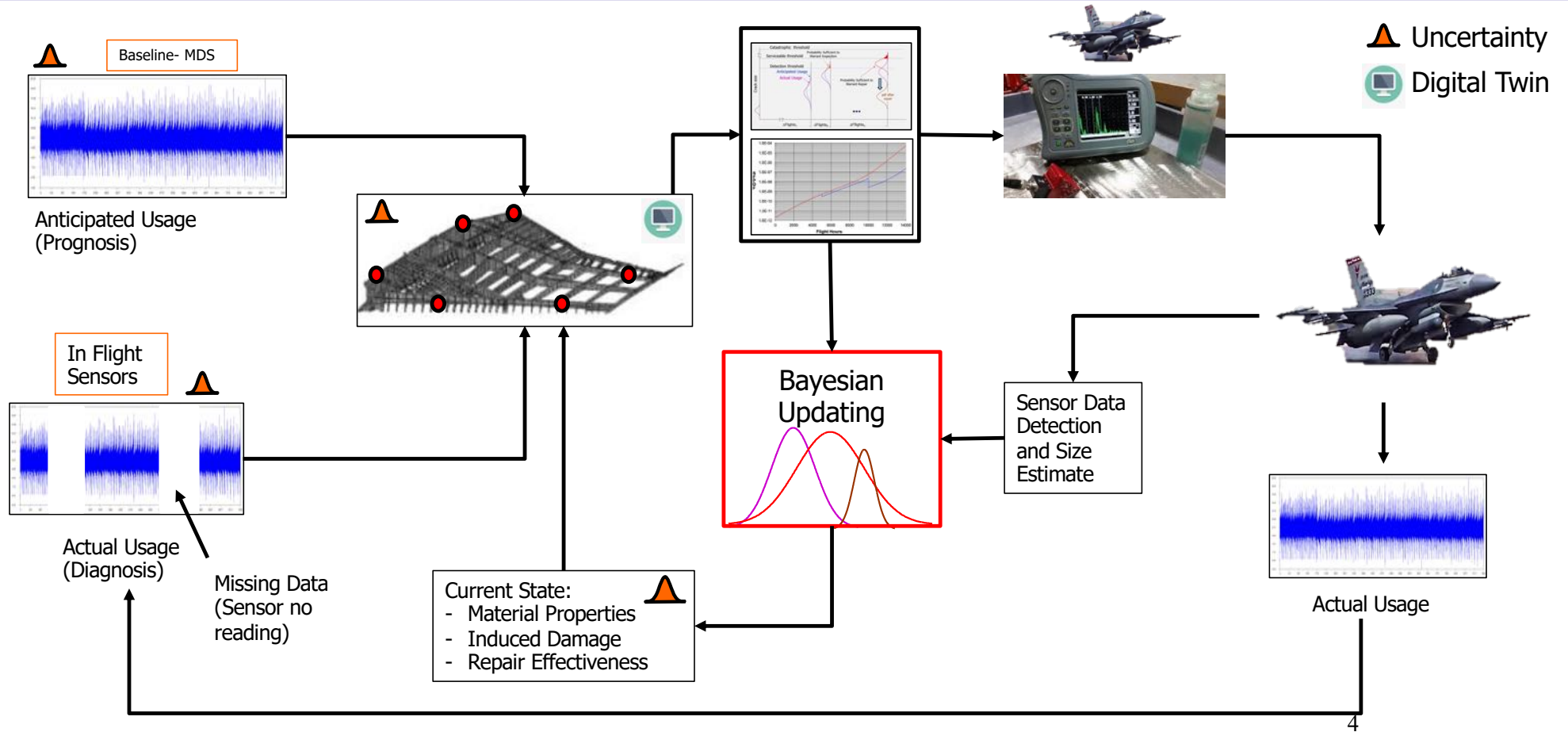


Motivation



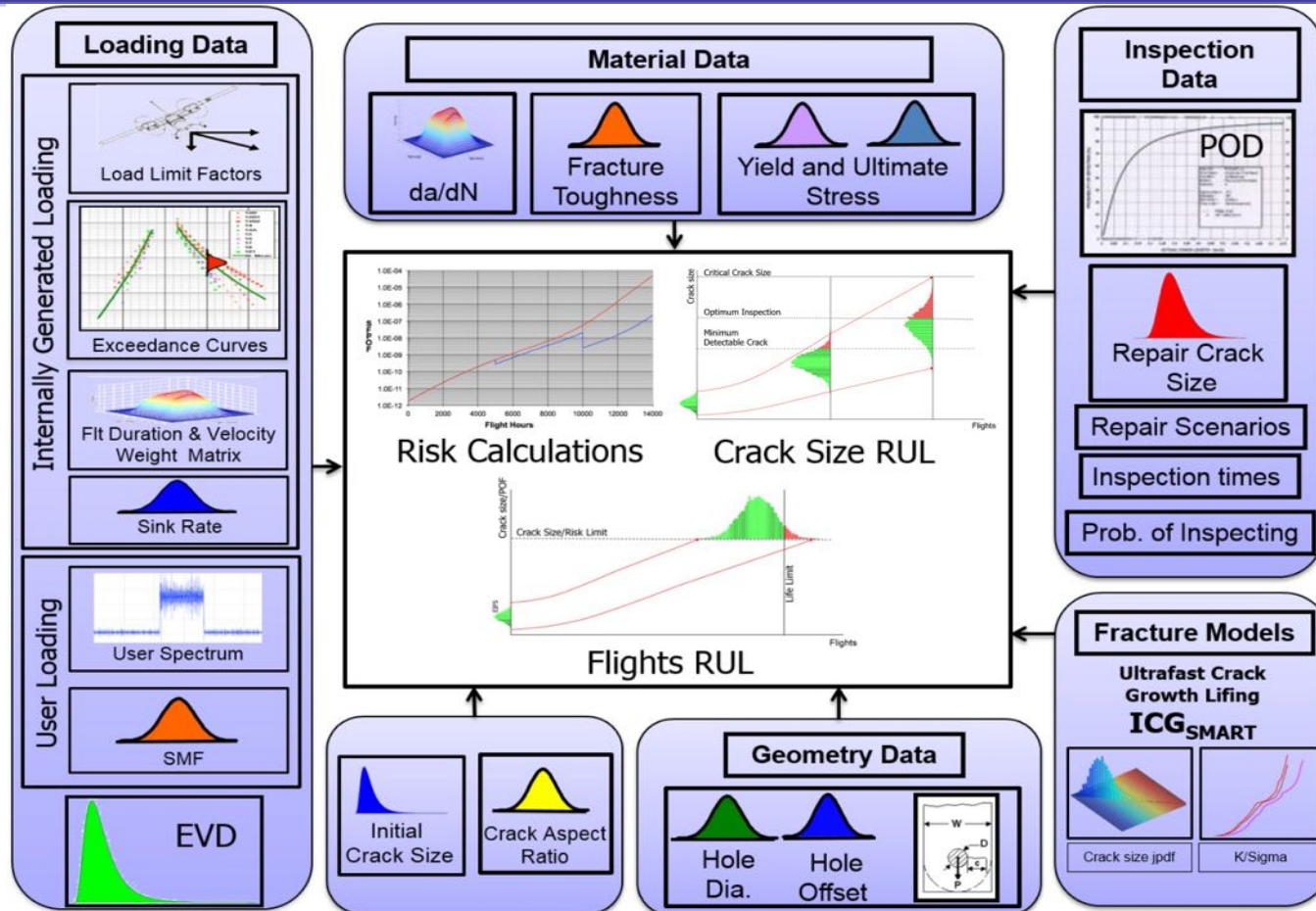


Motivation - Digital Twin





SMART|DT





Bayes theorem



$$\underbrace{P^+(\theta|\mathbf{D})}_{\text{Posterior distribution}} = \frac{\overbrace{L(\mathbf{D}|\theta)}^{\text{Likelihood}} \cdot \overbrace{P^-(\theta)}^{\text{Prior distribution}}}{\underbrace{\text{NF}}_{\text{Normalization factor}}}$$

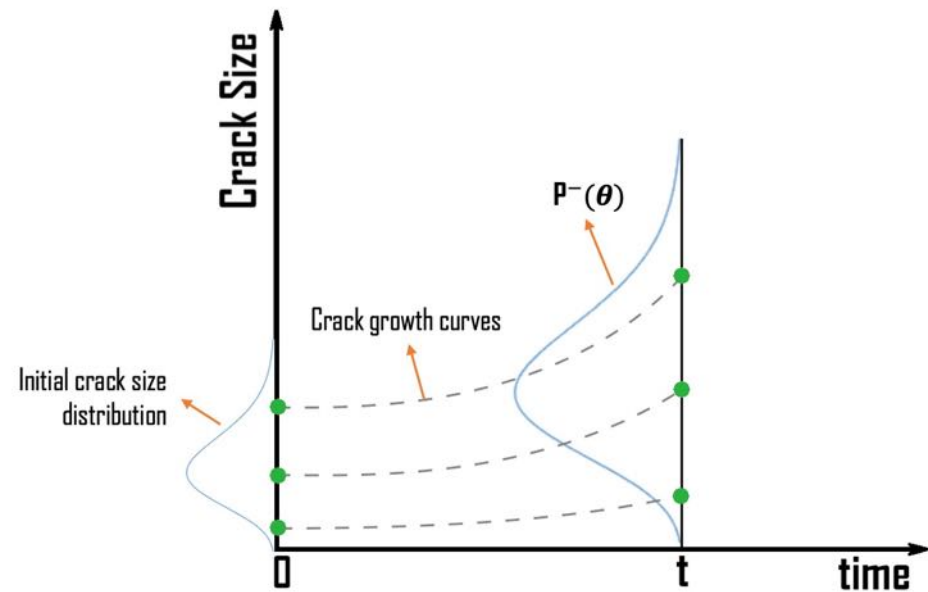
- θ represents the parameters mean(μ) \rightarrow independent variable and standard deviation(σ) \rightarrow assumed, it will be fixed,
- \mathbf{D} represents the vector of the measurements (or inspections),
- P^- represents the prior distribution \rightarrow Distribution of crack size at the time,
- $L(\mathbf{D}|\theta)$ represents the likelihood function of the parameters.
- **NF** Normalization Factor, used to get a probability density function.
- P^+ represents the posterior distribution given the detected crack sizes.



Bayesian Updating



$$\underbrace{P^+(\theta | \mathbf{D})}_{\text{Posterior distribution}} = \frac{\overbrace{L(\mathbf{D} | \theta)}^{\text{Likelihood}} \cdot \overbrace{P^-(\theta)}^{\text{Prior distribution}}}{\underset{\substack{\text{NF} \\ \downarrow \\ \text{Normalization factor}}}{}}$$

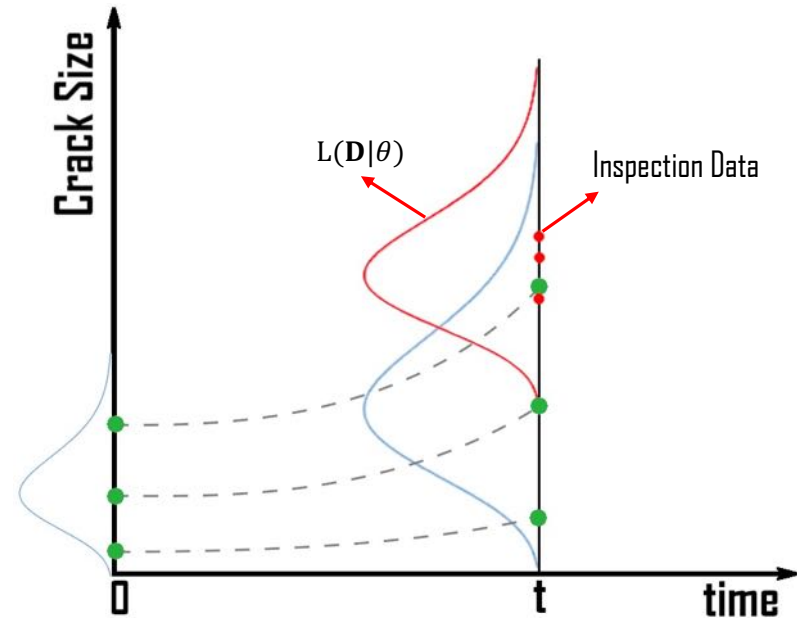




Bayesian Updating



$$\underbrace{P^+(\theta | \mathbf{D})}_{\text{Posterior distribution}} = \frac{\underbrace{L(\mathbf{D} | \theta)}_{\text{Likelihood}} \cdot \underbrace{P^-(\theta)}_{\text{Prior distribution}}}{\underbrace{\text{NF}}_{\text{Normalization factor}}}$$

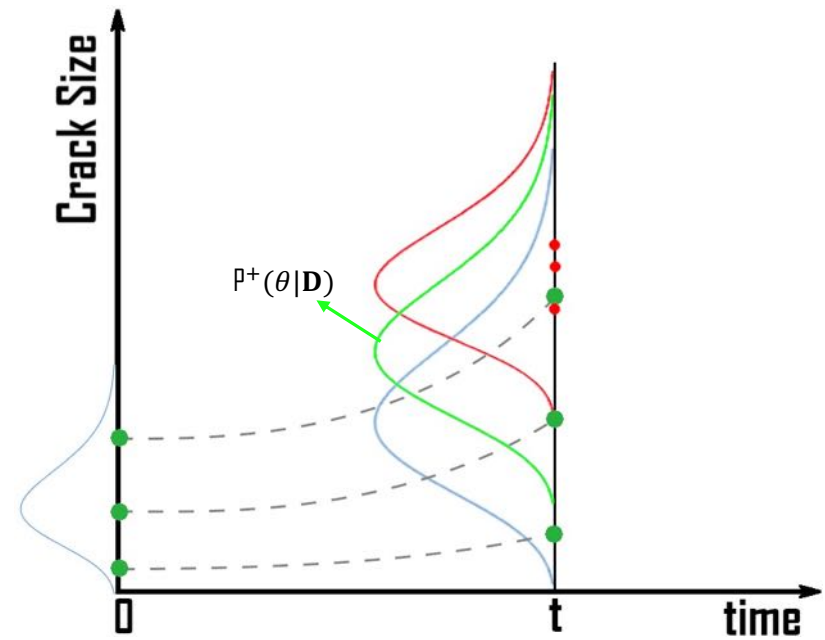




Bayesian Updating

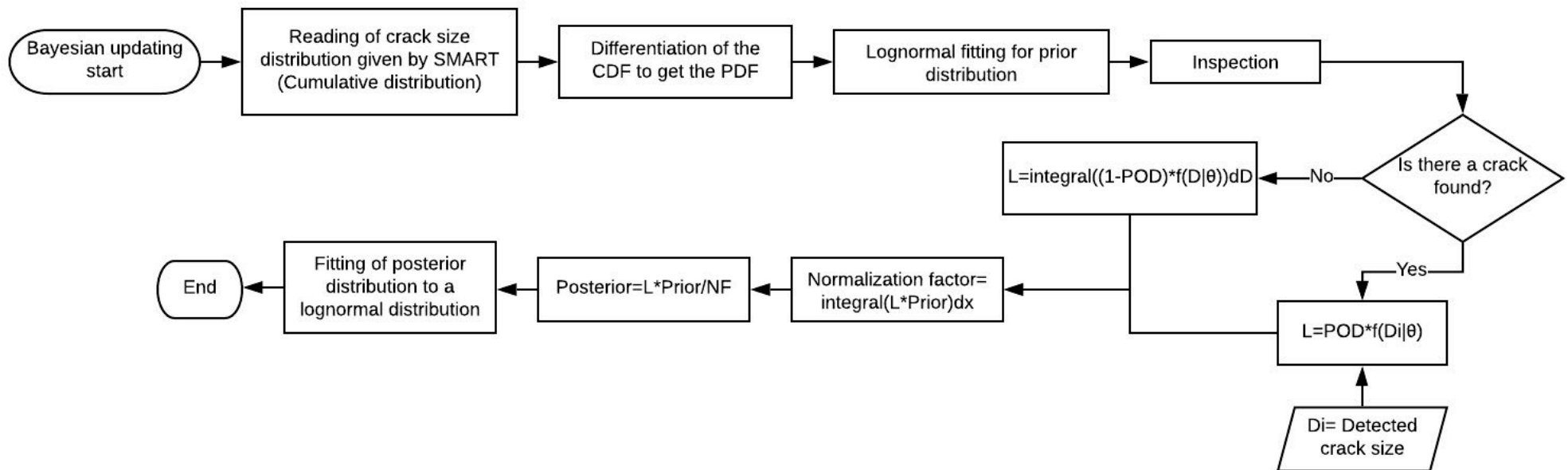


$$\underbrace{P^+(\theta | \mathbf{D})}_{\text{Posterior distribution}} = \frac{\underbrace{L(\mathbf{D} | \theta)}_{\text{Likelihood}} \cdot \underbrace{P^-(\theta)}_{\text{Prior distribution}}}{\underbrace{\text{NF}}_{\text{Normalization factor}}}$$





Flowchart



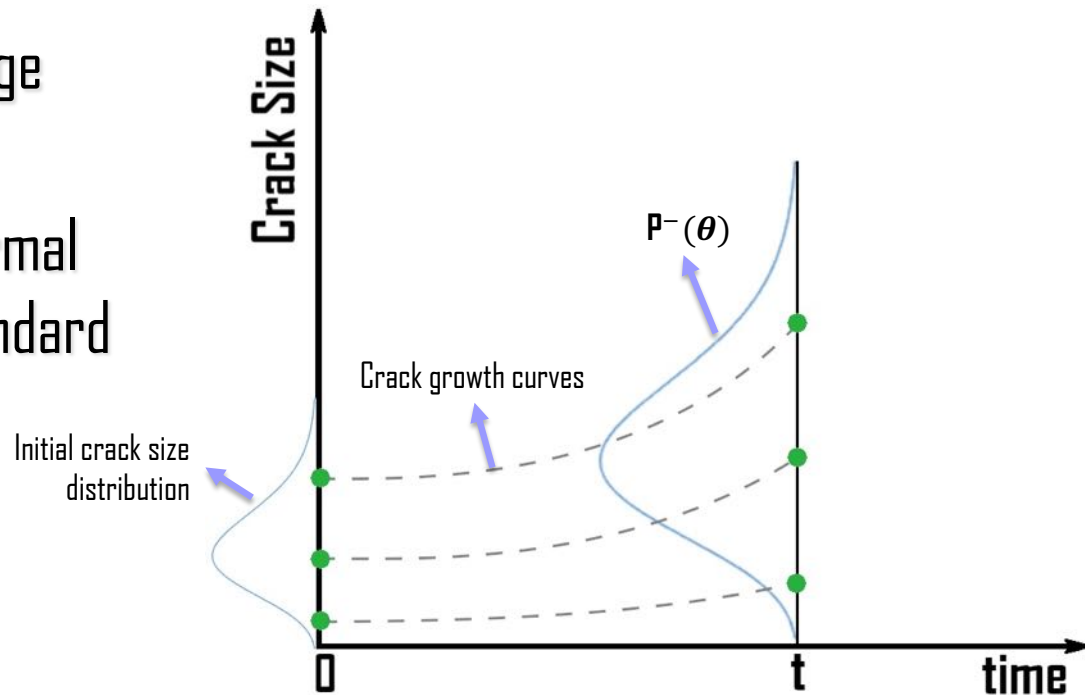
$f(D|\theta)$ Represents the distribution of means for the crack found D_i . 10



1. Prior distribution $P^-(\theta)$



- It is known, based on the damage tolerance model at time t .
- Assumed that follows a log-normal distribution with mean and standard deviation known.





2. Likelihood $L(\mathbf{D}|\theta)$



$$L(\mathbf{D}|\theta) = \underbrace{L_D(\theta)}_{\substack{\text{Likelihood} \\ \text{of Detection}}} \cdot \underbrace{L_{ND}(\theta)}_{\substack{\text{Likelihood} \\ \text{of NO} \\ \text{Detection}}}$$

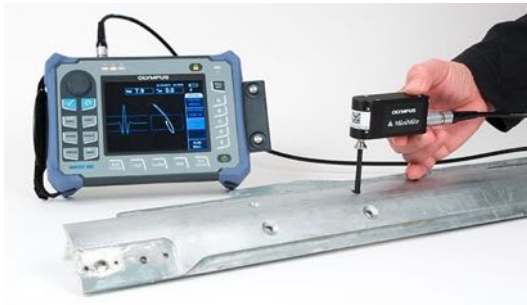
- The likelihood function reflects the degree of agreement between the obtained measurements, \mathbf{D} , and the output obtained from the mathematical model (Log-normal distribution) used to physically describe the system.
- It will be dependent on each inspection, and whether a crack is found or not.



Likelihood of Detection $L_D(\theta)$

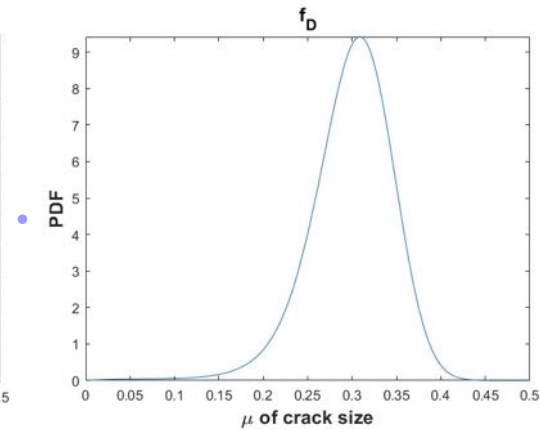
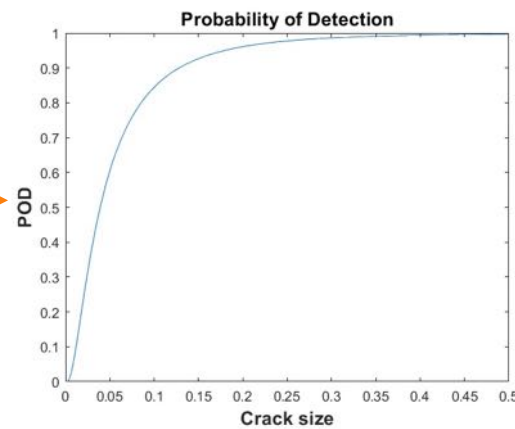


$$L_D(\theta) = \prod_{i=1}^{N_D} \underbrace{POD_i(D_i(t_i))}_{\text{Probability of Detection}} \cdot \underbrace{f(D_i(t_i)|\theta)}_{\text{PDF of crack size}}$$



Eddy current testing

Taken from: <https://www.olympus-ims.com/en/insight/ndt-level-3-eddy-current-testing/>



Represents the distribution of means for the crack found D_i .



Function $POD_i(D_i(t_i))$



- First, the probability of detection curve from the inspection method is used. It will follow a log-normal distribution with mean μ and standard deviation of σ

$$\text{LogNormal}(x|\varphi, h) = \frac{1}{x \cdot h \cdot \sqrt{2\pi}} \cdot \exp\left\{\frac{-(\log(x) - \varphi)^2}{2 \cdot h^2}\right\} \quad \text{for } x > 0$$

$$\varphi = \text{Log}\left(\frac{\mu^2}{\sqrt{\sigma^2 + \mu^2}}\right)$$

$$h = \sqrt{\text{Log}\left(\frac{\sigma^2}{\mu^2} + 1\right)}$$



Function $f(D_i(t_i))$



➤ It is defined as:

$$\text{LogNormal}(D|\varphi, h) = \frac{1}{D \cdot h \cdot \sqrt{2\pi}} \cdot \exp\left\{\frac{-(\log(D) - \varphi)^2}{2 \cdot h^2}\right\}, \text{ for } D > 0$$

$$\varphi = \text{Log}\left(\frac{\mu^2}{\sqrt{\sigma^2 + \mu^2}}\right), \text{ where } \mu \rightarrow \text{Independent variable and}$$
$$\sigma = \sigma_{\text{prior}}$$

$$h = \sqrt{\text{Log}\left(\frac{\sigma^2}{\mu^2} + 1\right)}$$



Likelihood for detected cracks



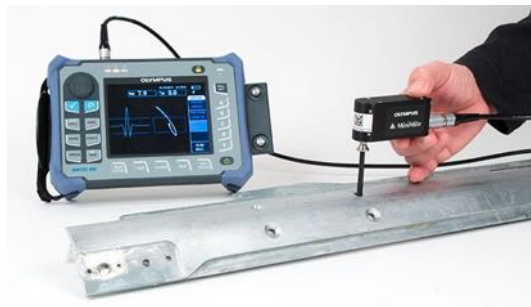
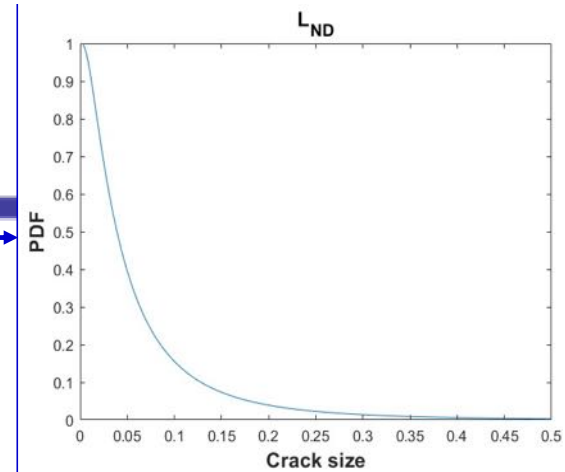
$$f(D_i(t_i) | \theta)$$

- Is a probability density function that represents the distribution of parameters for the crack detected.
- For this pdf, we know the crack size D , as random variable with mean D .
- The standard deviation can be:
 - Estimated as the mean squared error of $(D_k - M(\theta))$.
 - Set it as a fixed parameter based on prior calculations or knowledge.



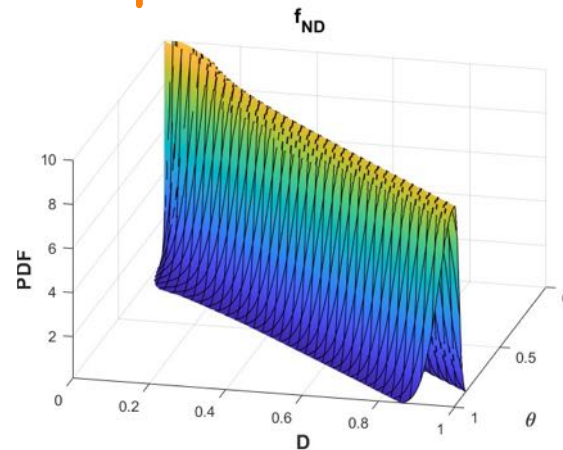
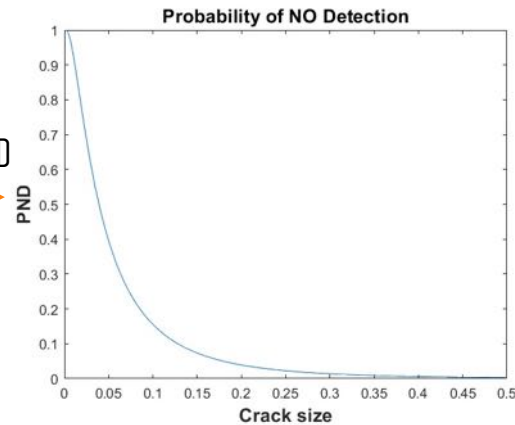
Likelihood of **NO** Detection $L_{ND}(\theta)$

$$L_{ND}(\theta) = \prod_{i=1}^{N_{ND}} \int_0^{\infty} \underbrace{PND_i(D(t_i))}_{\text{Probability of NO Detection}} \cdot \underbrace{f(D(t_i)|\theta)}_{\text{PDF}} dD$$



Eddy current testing

PND=1-POD



For the PDF we followed the same methodology than for detected data, but now we have D as random variable, The integral considers all the values D can take. 17

Taken from: <https://www.olympus-ims.com/en/insight/ndt-level-3-eddy-current-testing/>



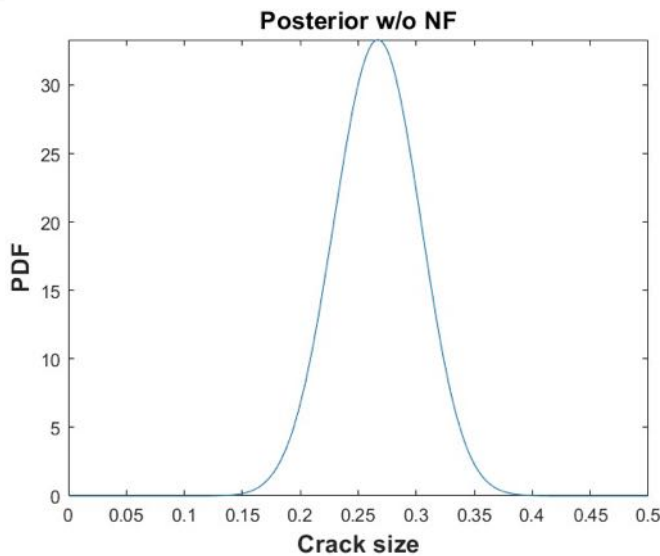
3. Normalization Factor



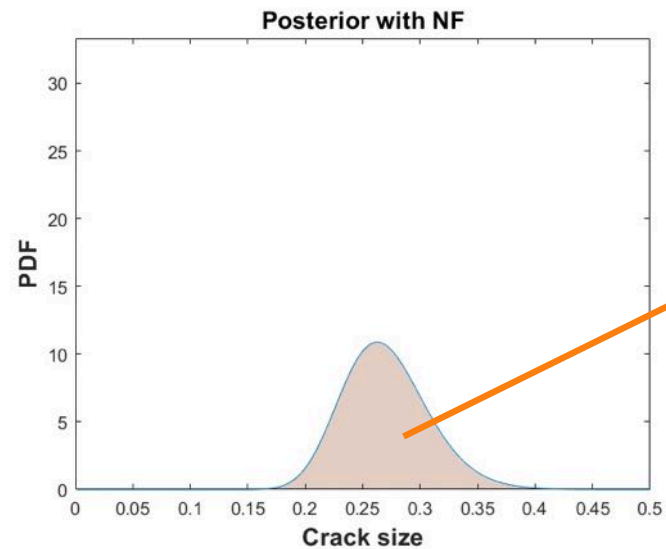
$$NF = \int_0^{\infty} \underbrace{L(\mathbf{D}|\theta)}_{\text{Likelihood}} \cdot \underbrace{P^-(\theta)}_{\text{Prior distribution}} \cdot d\theta$$

Likelihood Prior distribution

It's a normalization factor, so when we integrate the posterior distribution the cumulative density function is equal to 1.



NF →



$$\int_0^{\infty} P^+(\theta|\mathbf{D}) = 1$$

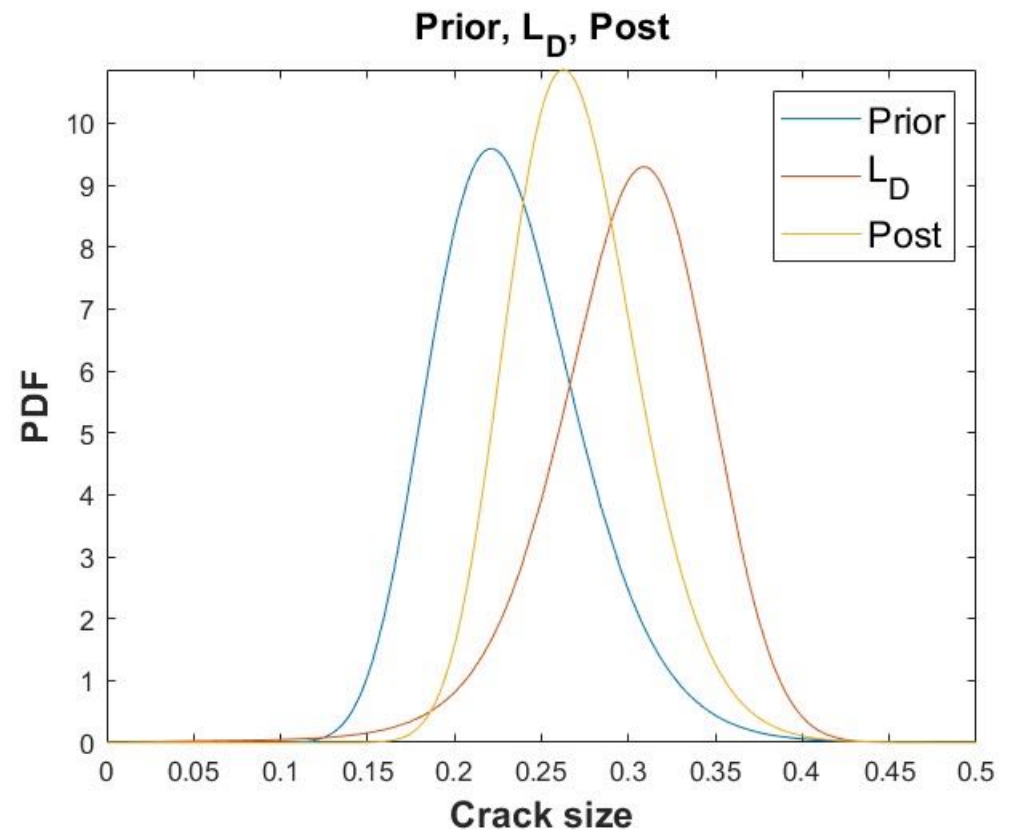


3. Posterior $P^+(\theta|\mathbf{D})$



- After computing the posterior distribution, we fit that expression to a log-normal distribution and get the new parameters.

$$P^+(\theta|\mathbf{D}) = \frac{L(\mathbf{D}|\theta) \cdot P^-(\theta)}{NF}$$



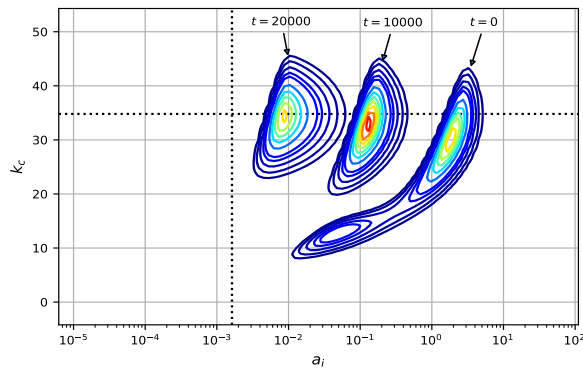


Multiple Importance Sampling Approach for PDTA

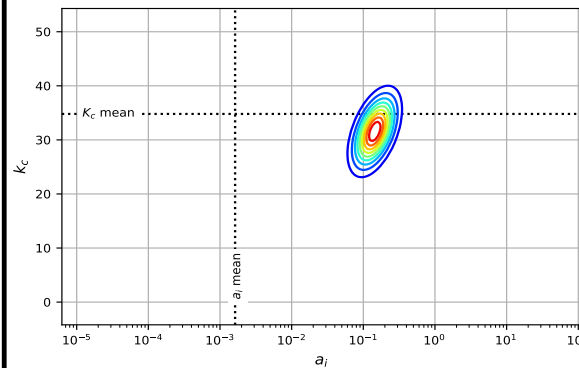


Individual Times

Important Region(s)

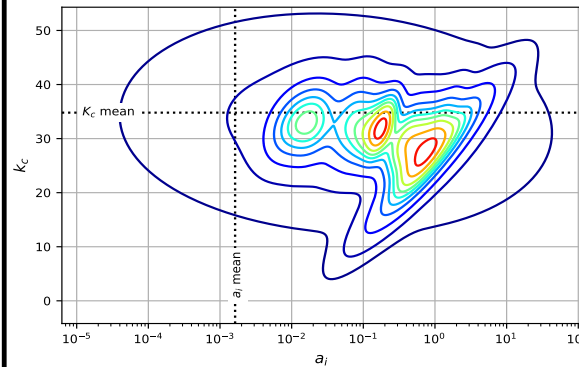
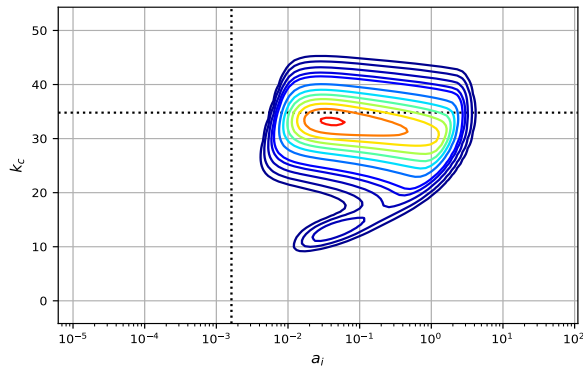


Adapted Sampling Region



- Basic Importance sampling
 - Adapt single sampling densities for individual evaluation times

Multiple Times



- Multiple Importance Sampling
 - Adapt a mixture density for a range of evaluation times

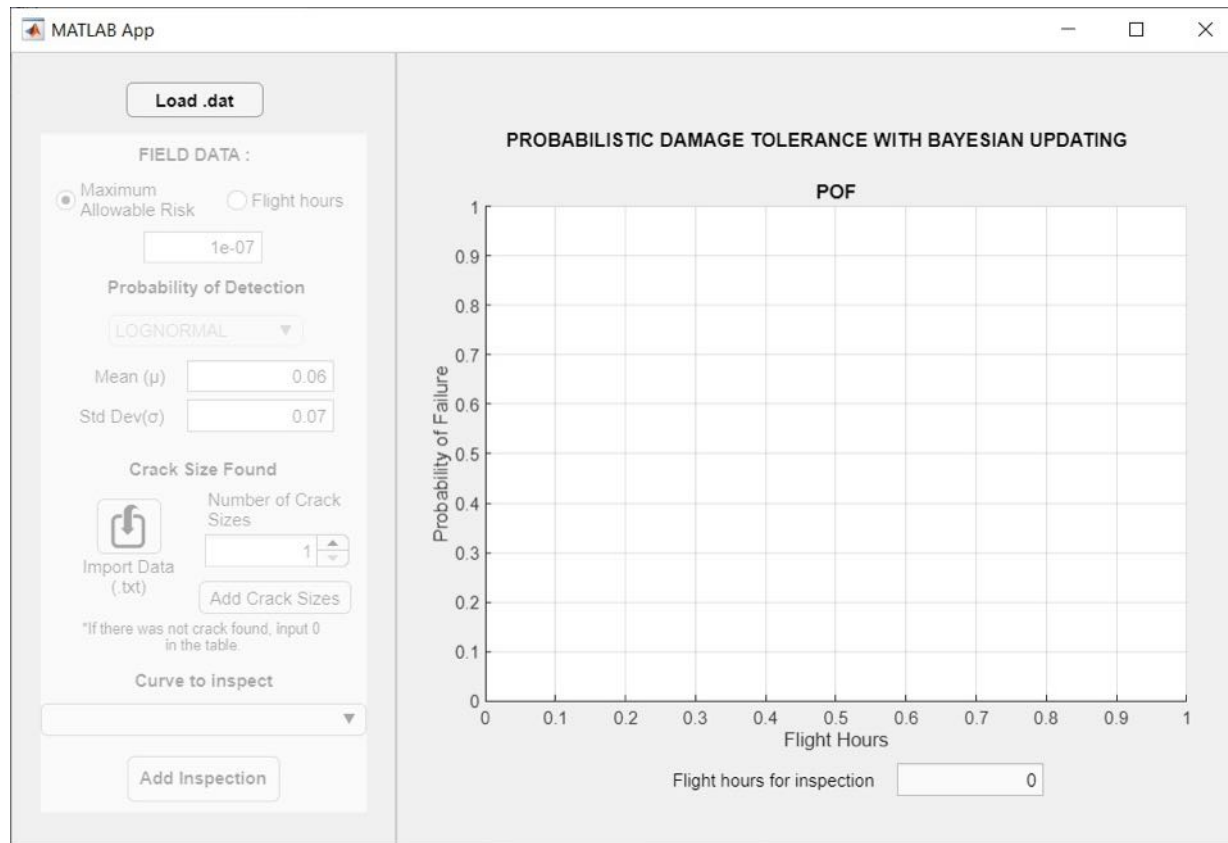
The PDTA AMIS algorithm estimates POF for PDTA using 6 orders of magnitude fewer samples compared to SMC for probabilities of 10^{-7}



MATLAB Script



Program interface





1. Reading .dat



The screenshot illustrates the workflow in MATLAB for reading a .dat file and generating a Probabilistic Damage Tolerance (POF) plot. It includes a file selection dialog, a code editor with aircraft parameters, and two side-by-side plots comparing standard and Bayesian-updated results.

Select File to Open Dialog: Shows the file 'Test.dat' (2 KB, DAT File) selected in the 'Final Program > V3' directory.

Code Editor (Aircraft Parameters):

```
1 | AIRCRAFT INFORMATION
2 |
3 | TITLE = Wing_Spar
4 | AC_NAME = Acme
5 | AC_MODEL = Sky Runner
6 | AC_SERIAL_NUM = SR100
7 | AC_TCS = TCSR100
8 |
9 | METHOD
10 |
11 |
12 |
13 | INTEGRATION_METHOD = MC 1000000 2394
14 | POF_PAK_ID = 40000 400
15 | ANALYSIS_TIME_UNITS = flights
16 |
17 | FRACTURE MECHANICS
18 | CRACK_GROWTH_CODE = MASTER_USER MasterCurve_e1000.avsn
19 | INITIAL_CRACK_SIZE = LOGNORMAL 0.005 0.003
20 | FRACTURE_TOUGHNESS = NORMAL 34.9 3.4
21 | YIELD_STRENGTH = DETERMINISTIC 120.0
22 |
23 | INSPECTIONS
24 | INSPECTIONS = 0
25 |
26 |
27 | LOADING_AND_ENV_PARAMETERS
```

Left Plot: PROBABILISTIC DAMAGE TOLERANCE

FIELD DATA: Maximum Allowable Risk (selected), Flight hours, Probability of Detection: LOGNORMAL, Mean (μ): 0.06, Std Dev(σ): 0.07, Crack Size Found: 1.

Y-axis: Probability of Failure (0 to 1). X-axis: Flight Hours (0 to 1).

Curve to inspect: Uninspected.

Flight hours for inspection: 0.

Right Plot: PROBABILISTIC DAMAGE TOLERANCE WITH BAYESIAN UPDATING

FIELD DATA: Maximum Allowable Risk (selected), Flight hours, Probability of Detection: LOGNORMAL, Mean (μ): 0.06, Std Dev(σ): 0.07, Crack Size Found: 1.

Y-axis: Probability of Failure (log scale, 10^{-20} to 10^0). X-axis: Flight Hours (0 to 4 $\times 10^4$).

Curve to inspect: Uninspected.

Flight hours for inspection: 0.



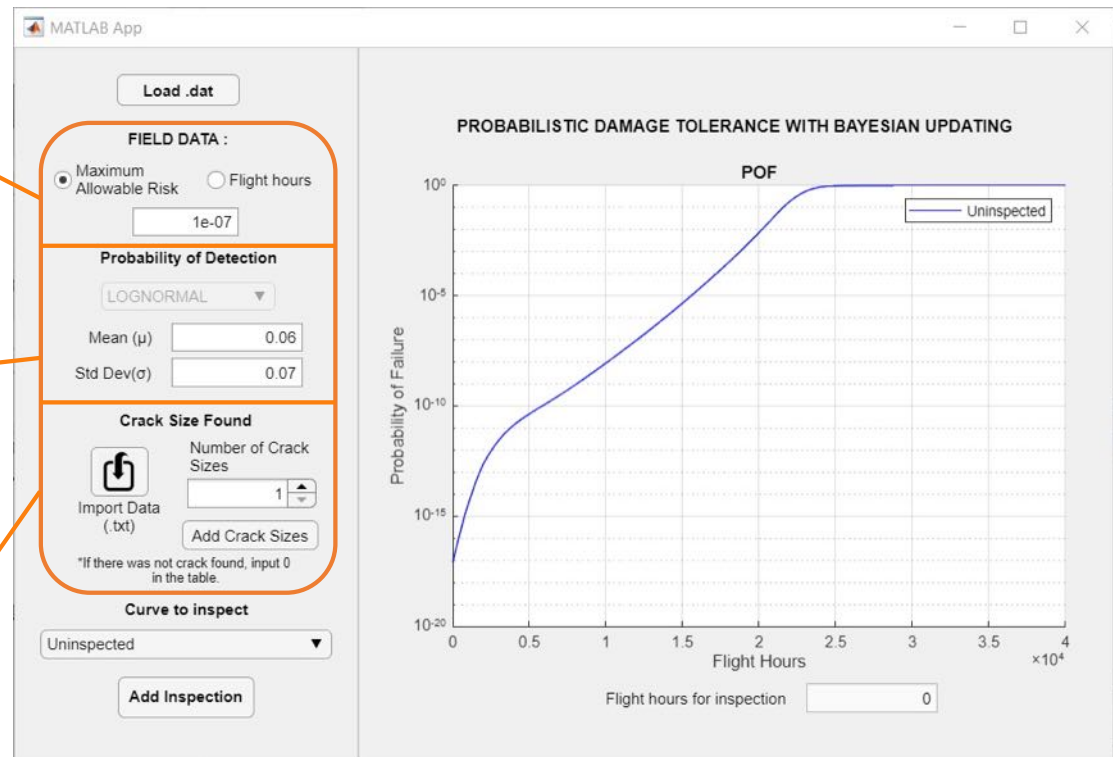
2. Input Field Data



Choose the risk or time to perform an inspection

Input the Probability of Detection (POD) parameters from the inspection method

Two options to add crack sizes: Importing a txt file or manually inputting them. (No limit on the number of crack sizes)





2.1 Crack Size Found During Inspection



• Importing

FIELD DATA :

Maximum Allowable Risk Flight hours

1e-07

Probability of Detection

LOGNORMAL

Mean (μ) 0.06

Std Dev(σ) 0.07

Crack Size Found

Import Data (.txt) Add Crack Sizes

*If there was not crack found, input 0 in the table.

Curve to inspect

Uninspected

Add Inspection

Flight hours for inspection 0

• Manual Input

FIELD DATA :

Maximum Allowable Risk Flight hours

1e-07

Probability of Detection

LOGNORMAL

Mean (μ) 0.06

Std Dev(σ) 0.07

Crack Size Found

Manual Input Add Crack Sizes

Number of Crack Sizes 3

*If there was not crack found, input 0 in the table.

Curve to inspect

Uninspected

Add Inspection

Flight hours for inspection 0

Crack_Size	Value
Crack_Size	0.3000
	0.2500
	0.2000

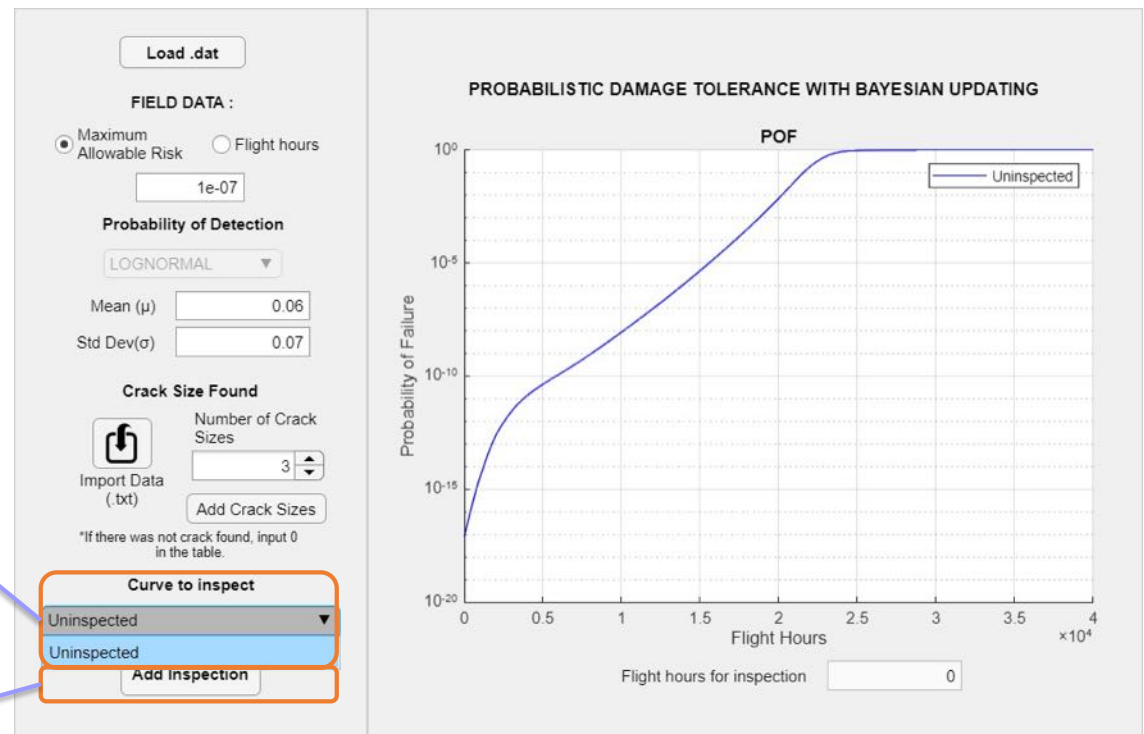


2.2. Select a PDF curve in which the Bayesian is performed



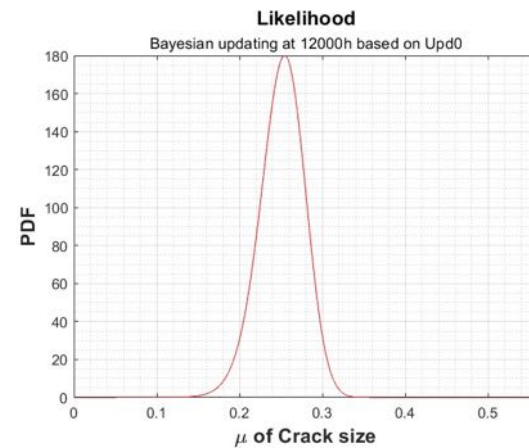
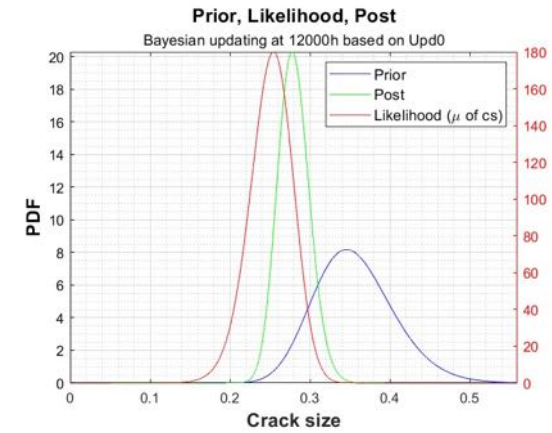
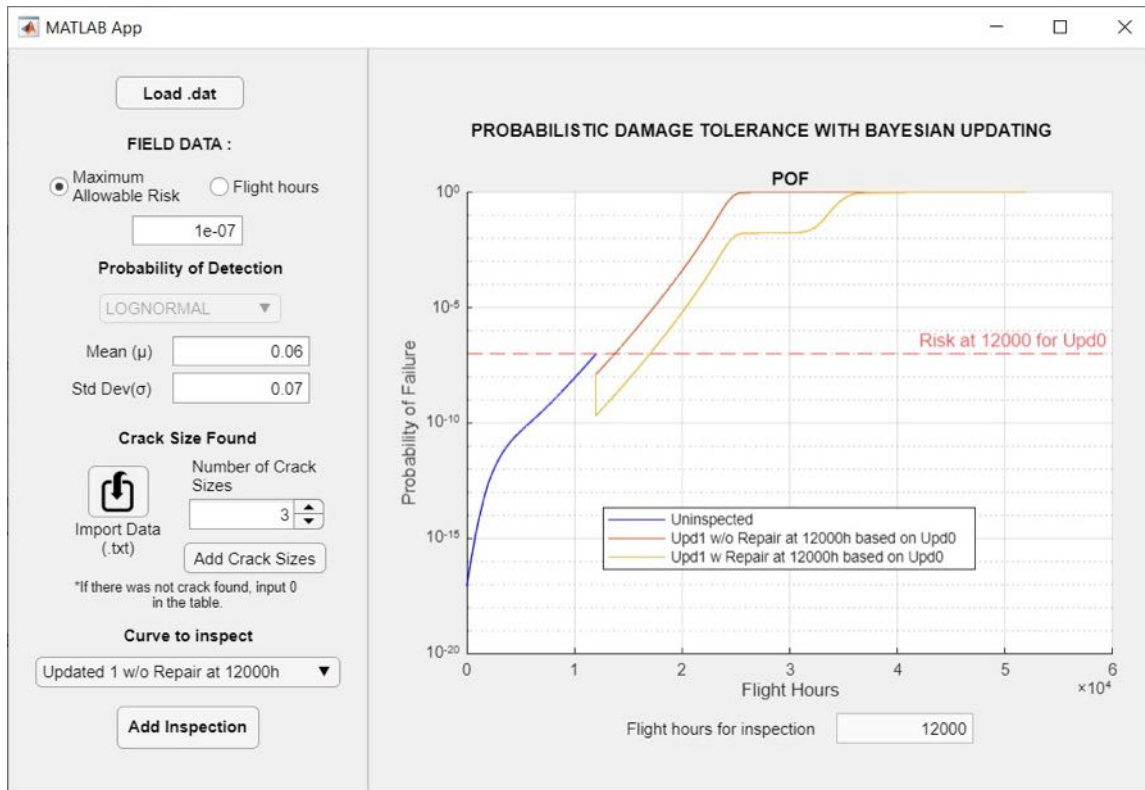
Select a PDF curve in which the Bayesian updating inspection will be performed.

Click "Add Inspection" to perform Bayesian Updating





3. Result of inspection added



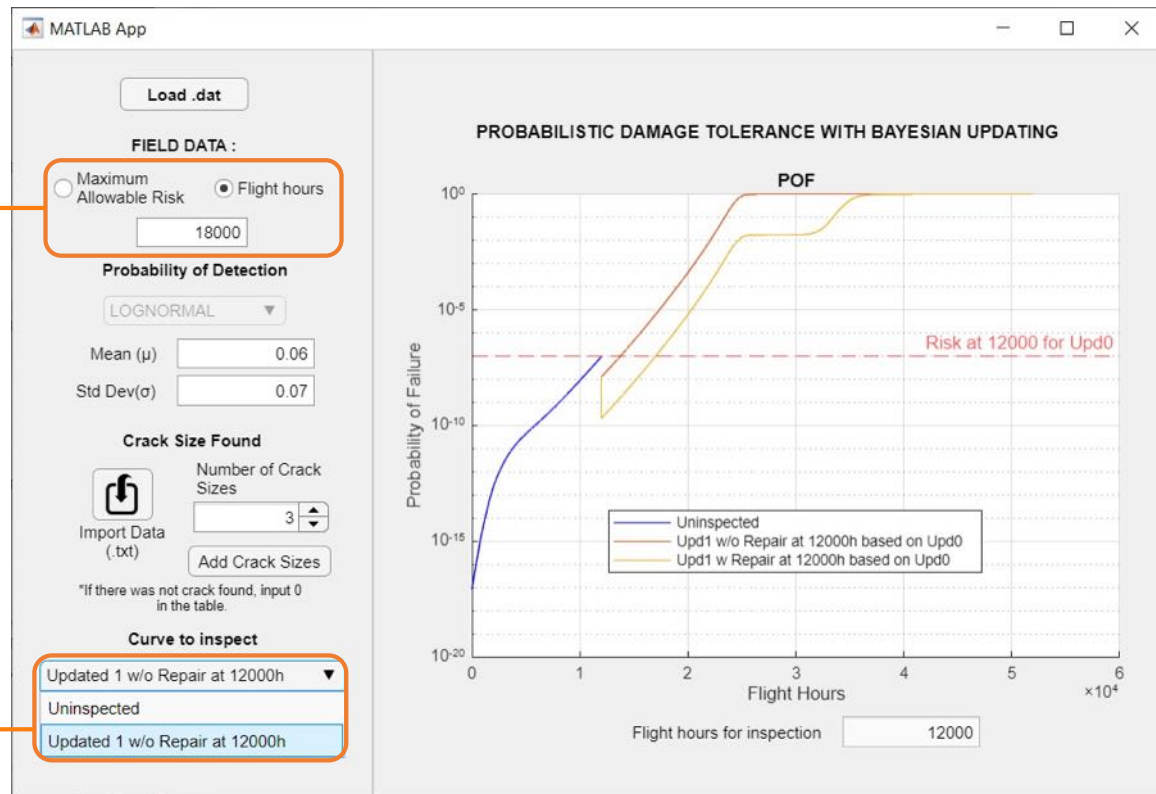


4. New Inspection



With Flight hours:
18000

Curve to Inspect:
Updated I





5. Results of new inspection



Load .dat

FIELD DATA :

Maximum Allowable Risk Flight hours

18000

Probability of Detection

LOGNORMAL

Mean (μ) 0.06

Std Dev(σ) 0.07

Crack Size Found

Number of Crack Sizes

3

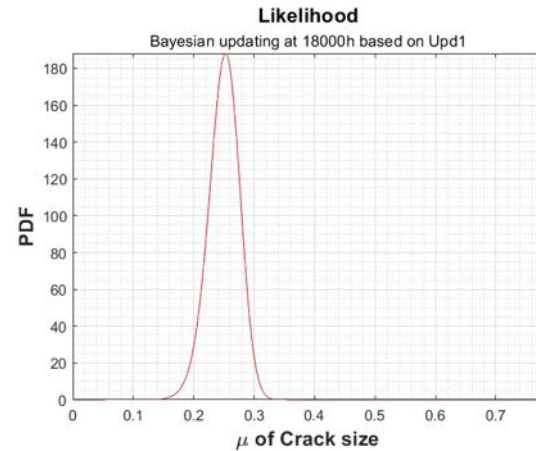
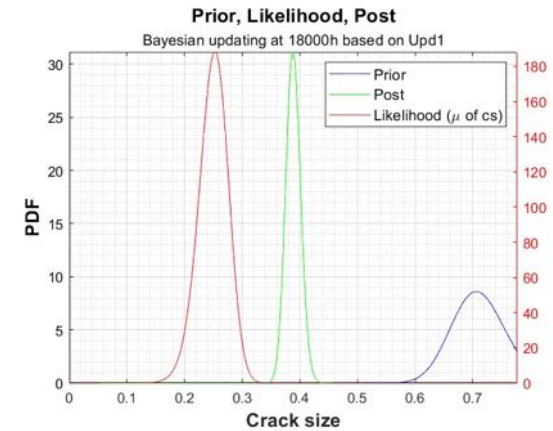
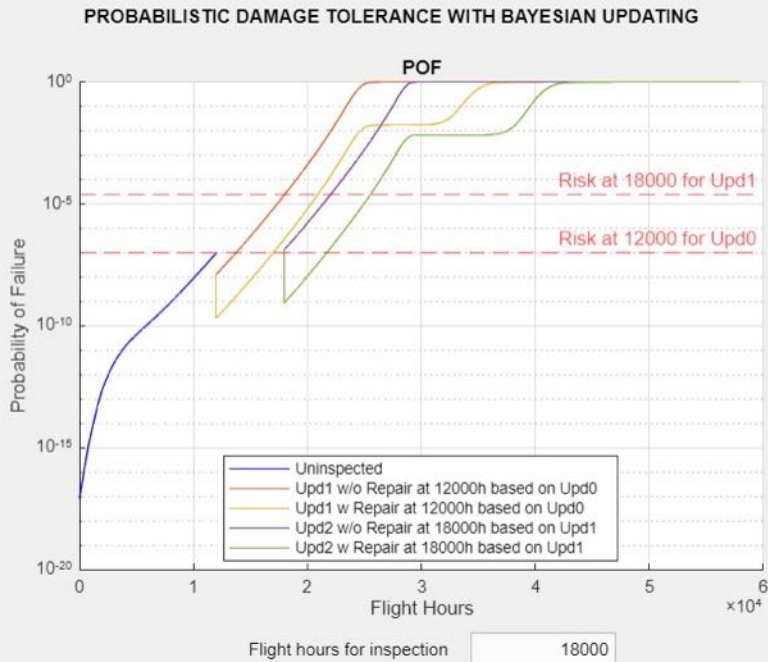
Import Data (.txt) Add Crack Sizes

*If there was not crack found, input 0 in the table.

Curve to inspect

Updated 2 w/o Repair at 18000h

Add Inspection





EXAMPLES



		INSPECTION			
		1 AIRPLANE	2 AIRPLANES		
1	D		D-D	3	
2	ND		D-ND	4	
			ND-ND	5	

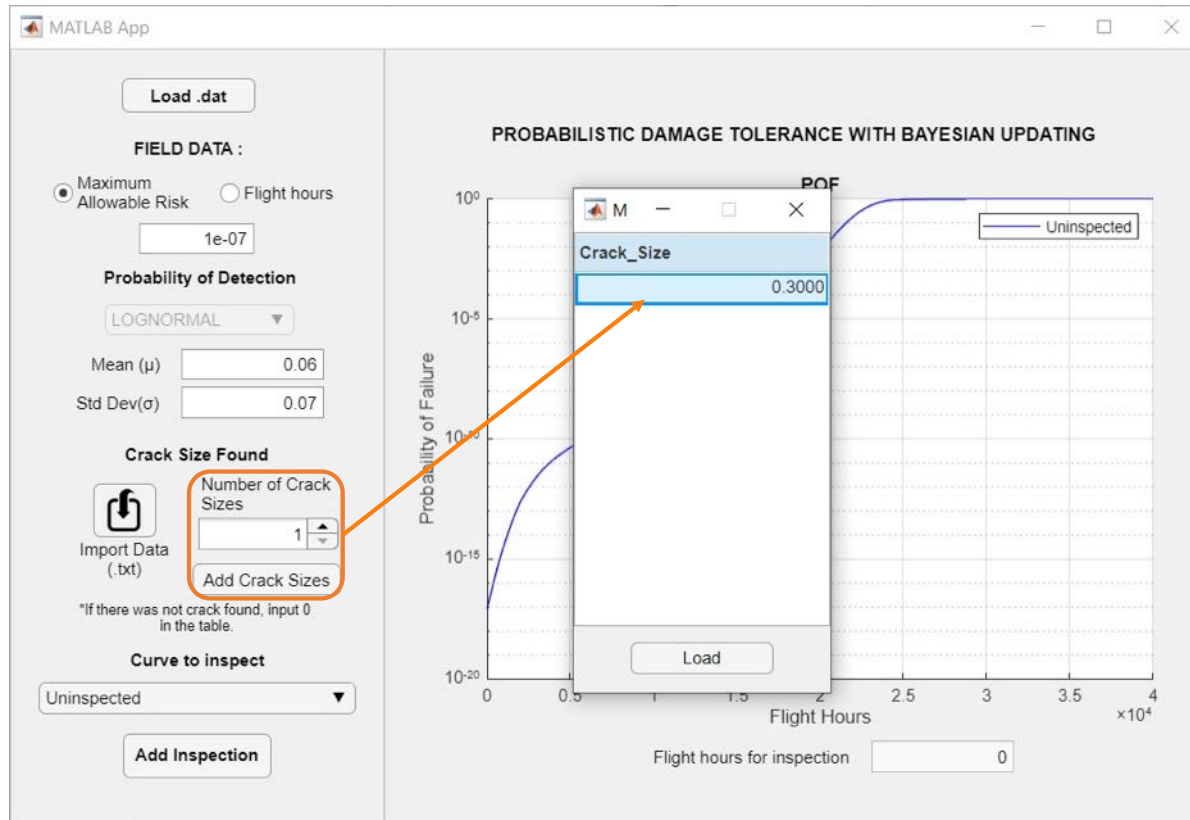
- 1 Detection = 1 crack in 1 airplane = D_i
- D=Detection
- ND= No Detection



Crack size detection = 0.3 in.

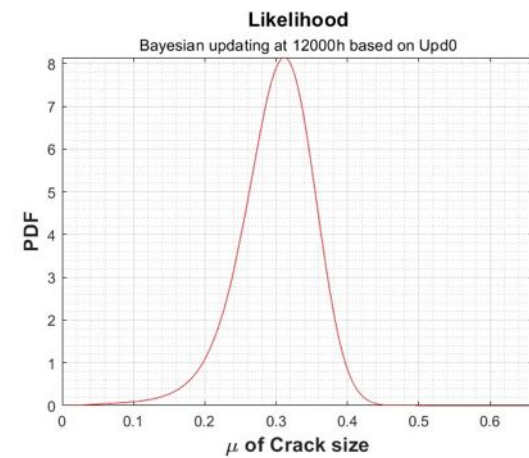
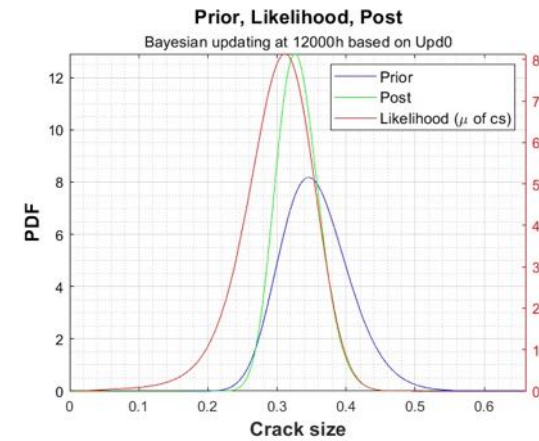
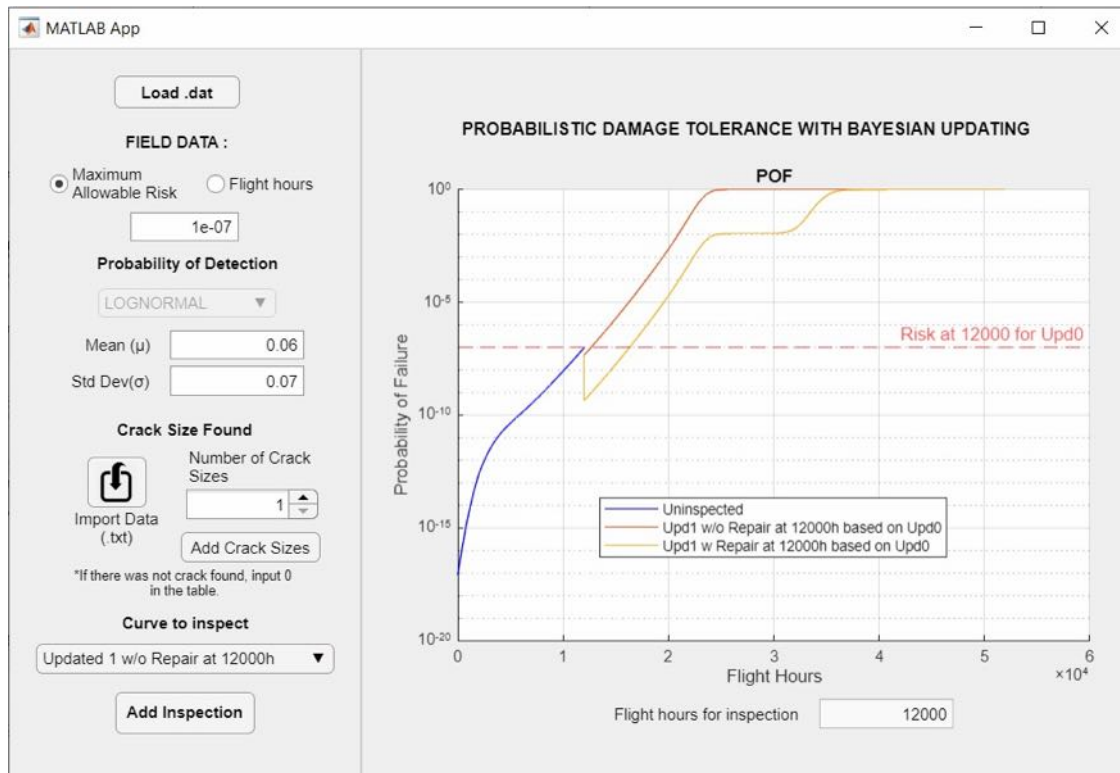


1





Results Crack size detection = 0.3 in.

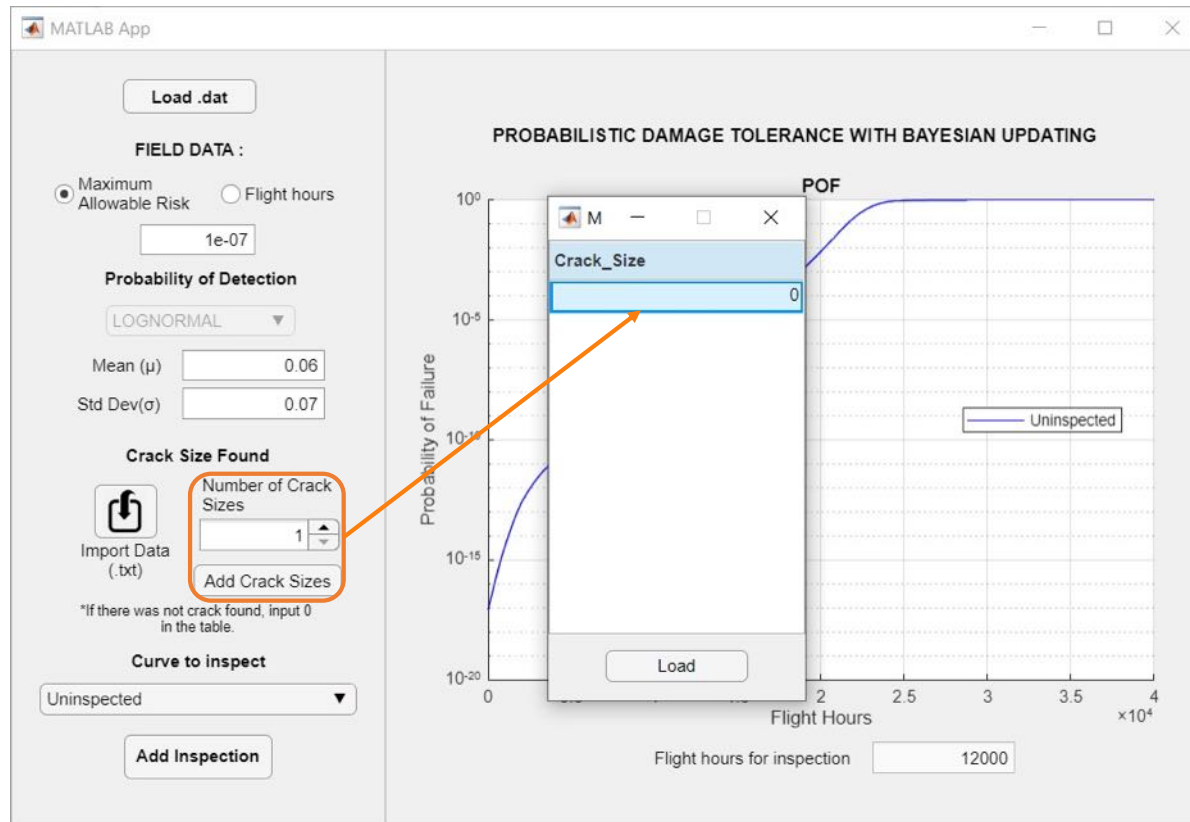




No detection

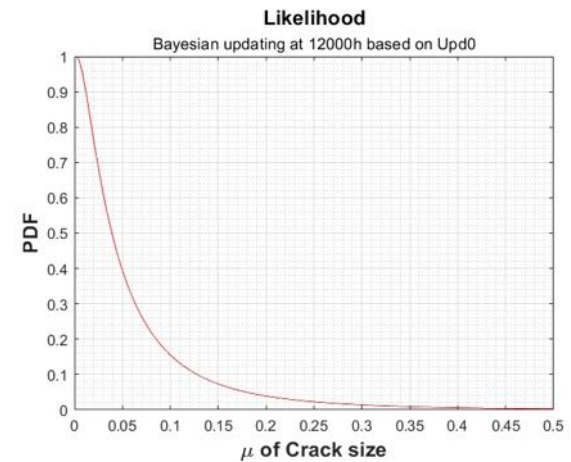
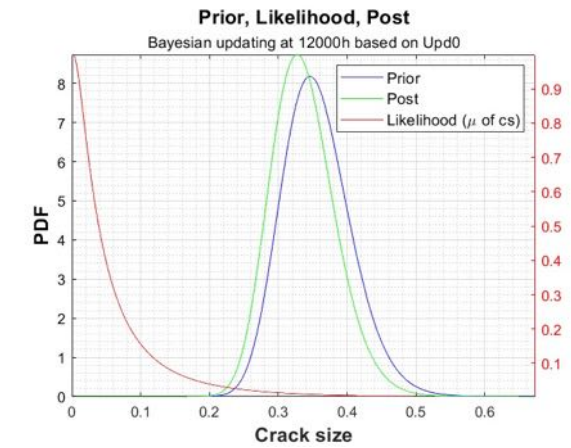
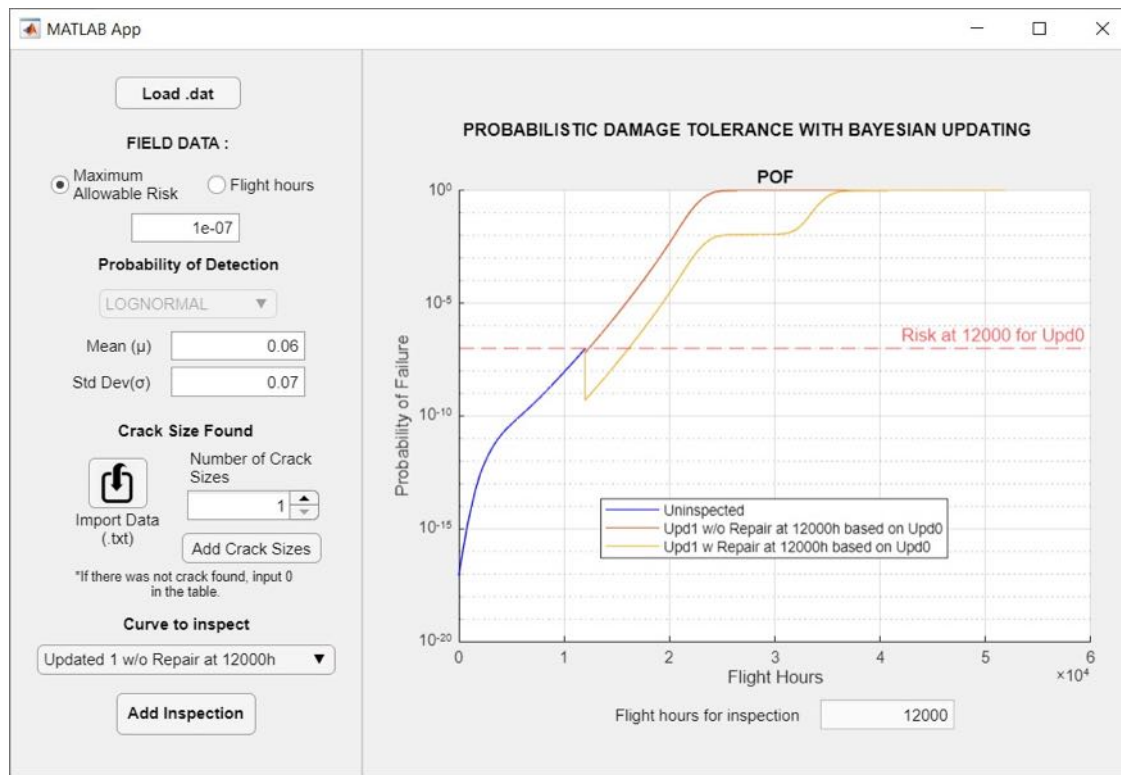


2





Results Crack size No detection

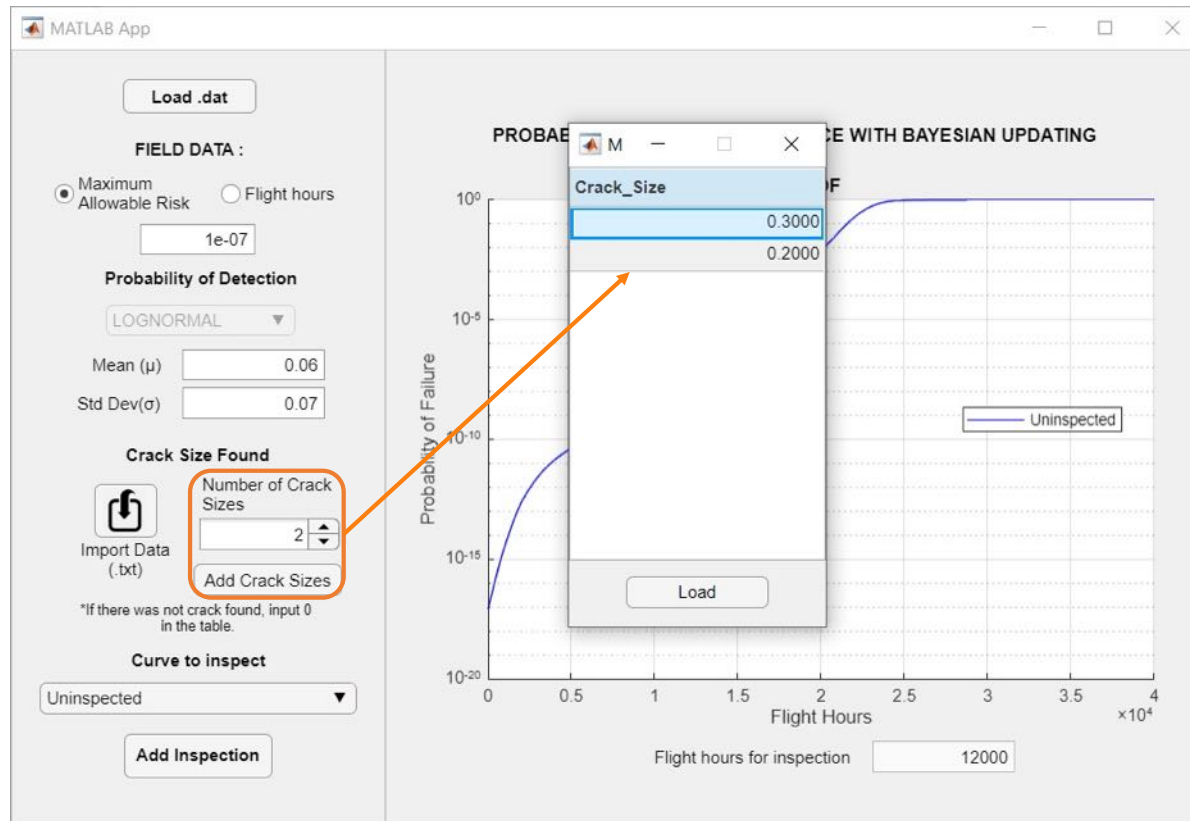




Crack size detections = 0.3 and 0.2 in.

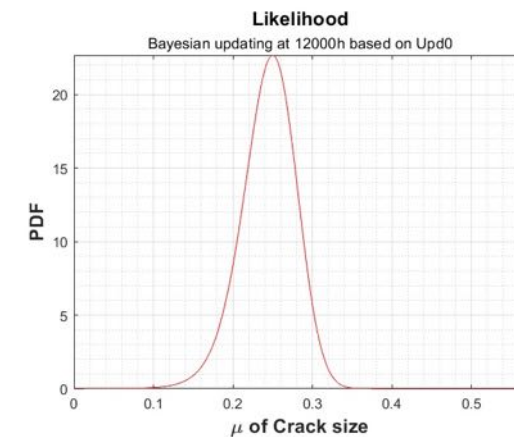
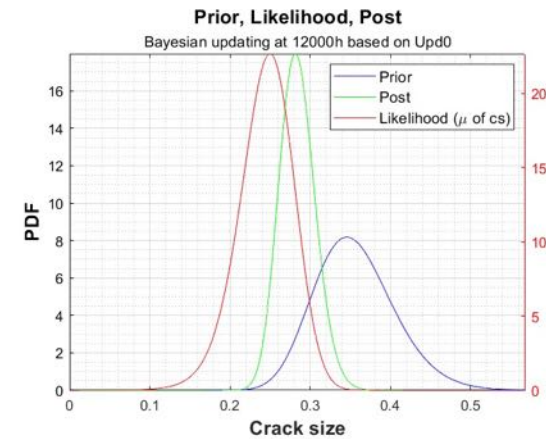
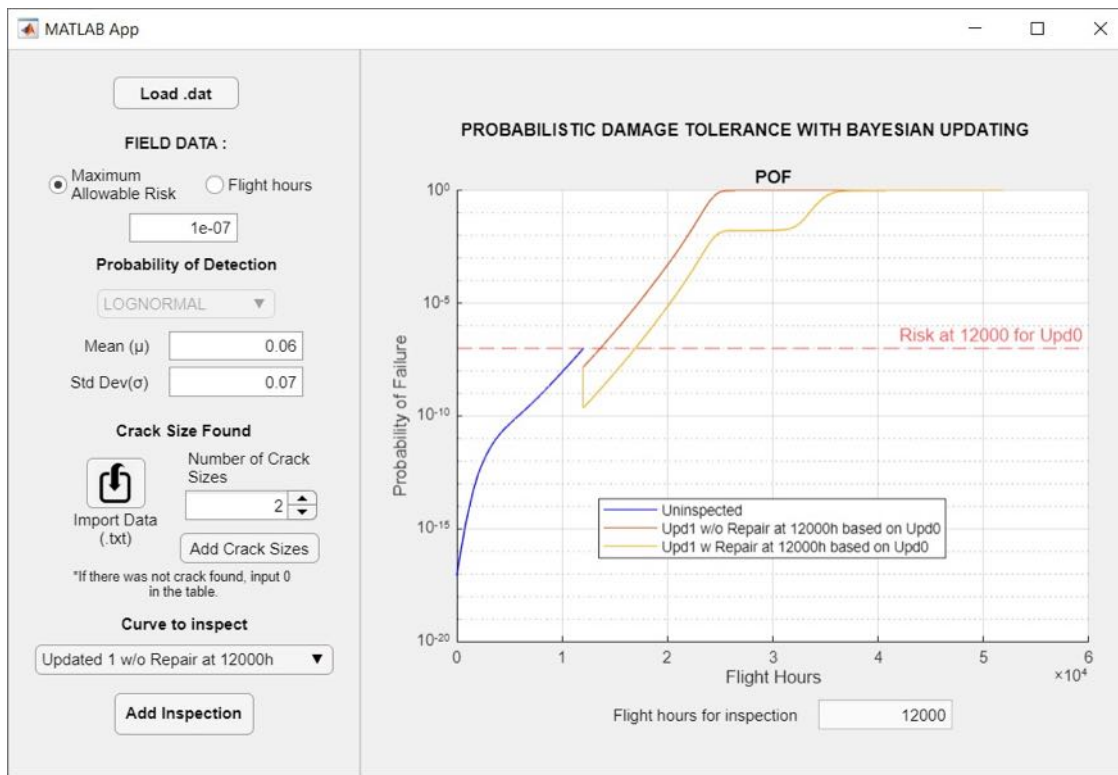


3





Results Crack size det. = 0.3 and 0.2 in

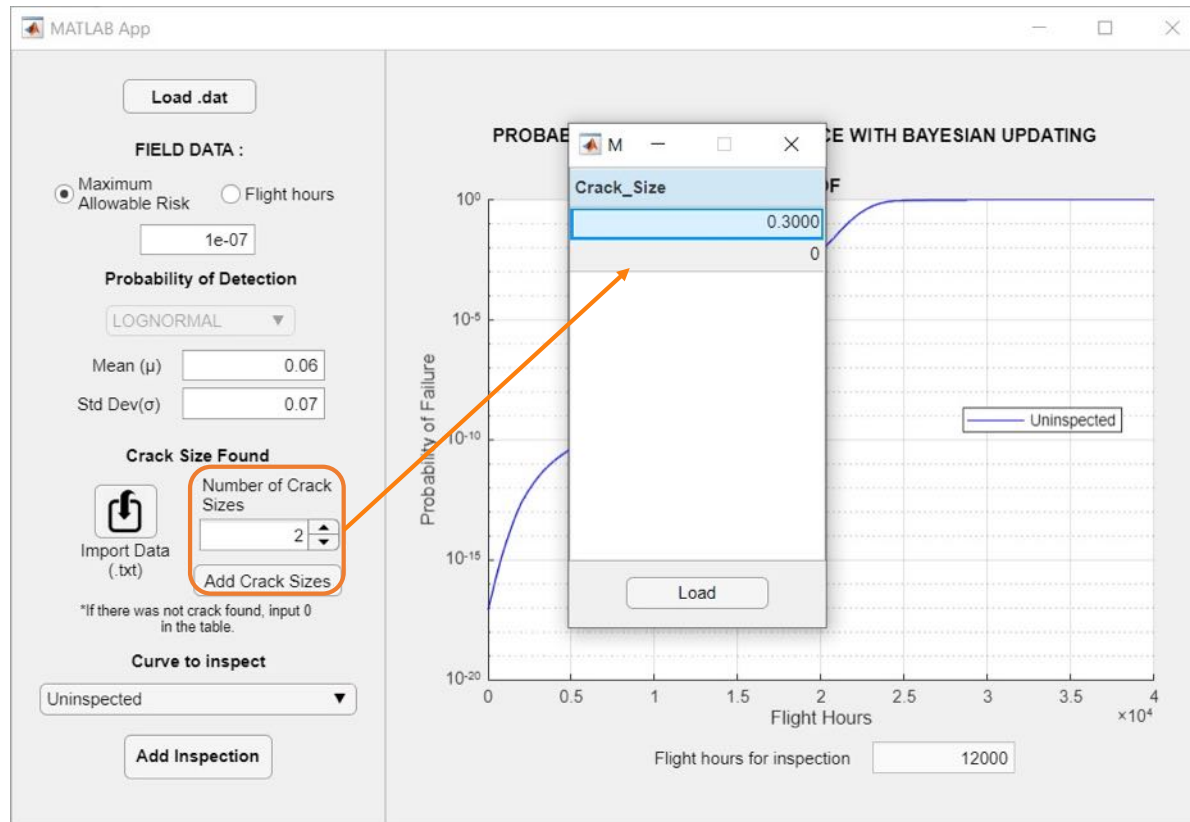




Crack size detections = 0.3 in. and one no detection

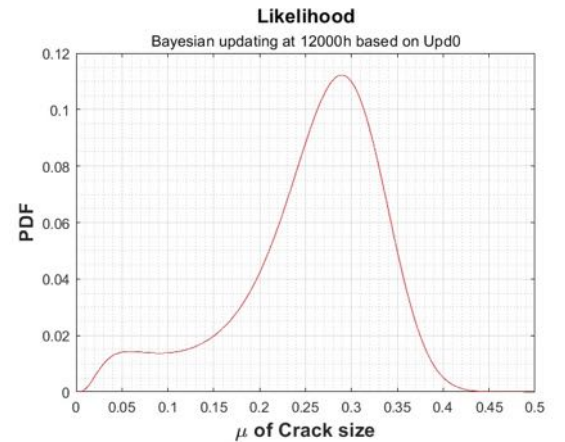
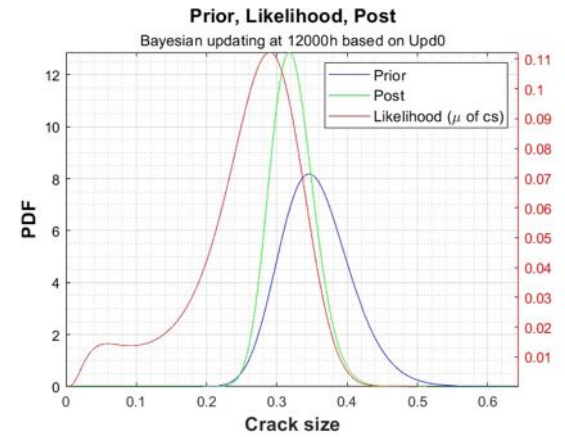
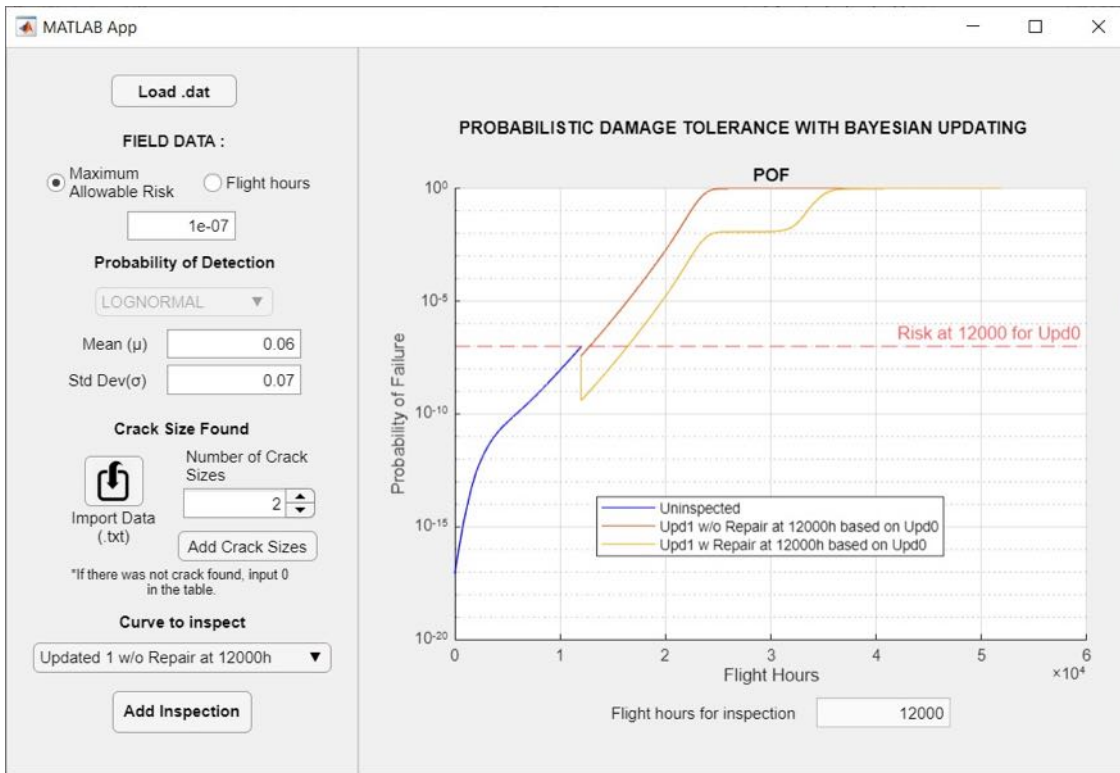


4





Results Crack size det.= 0.3 in. and ND

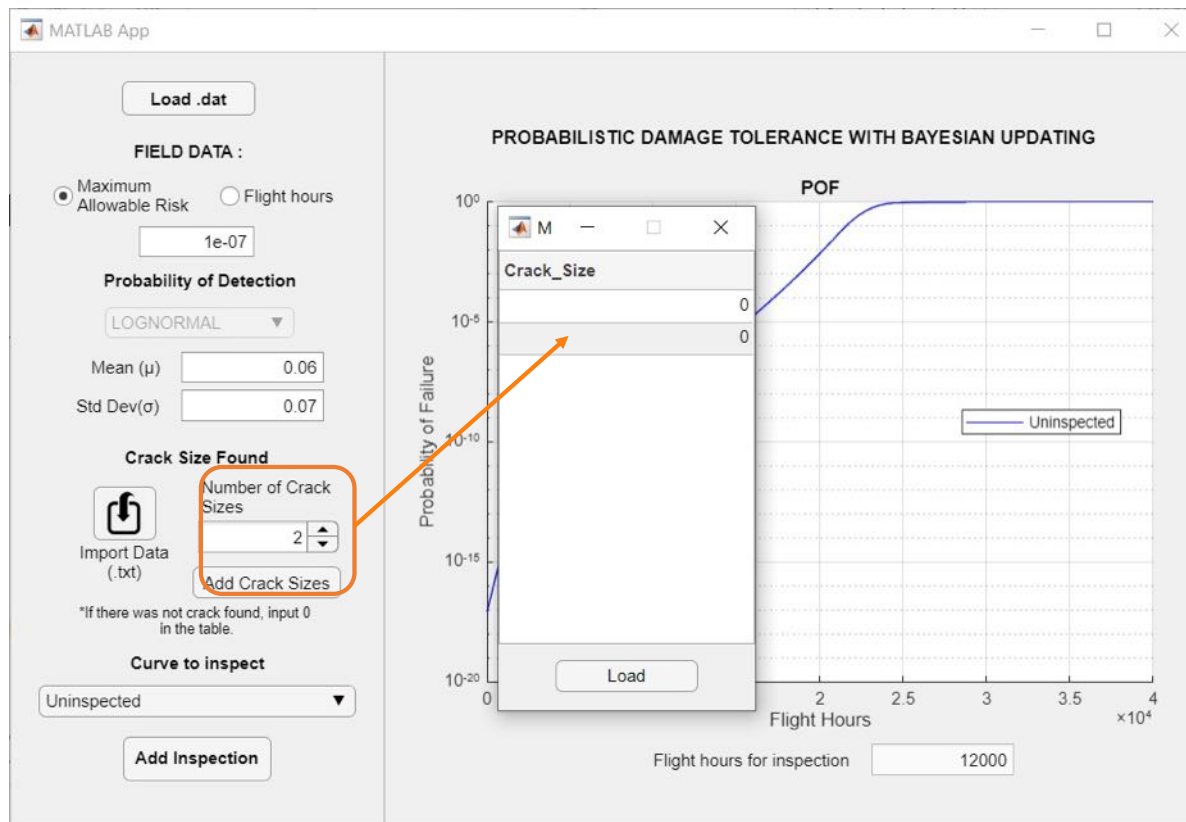




Posterior for two inspections and no detected cracks

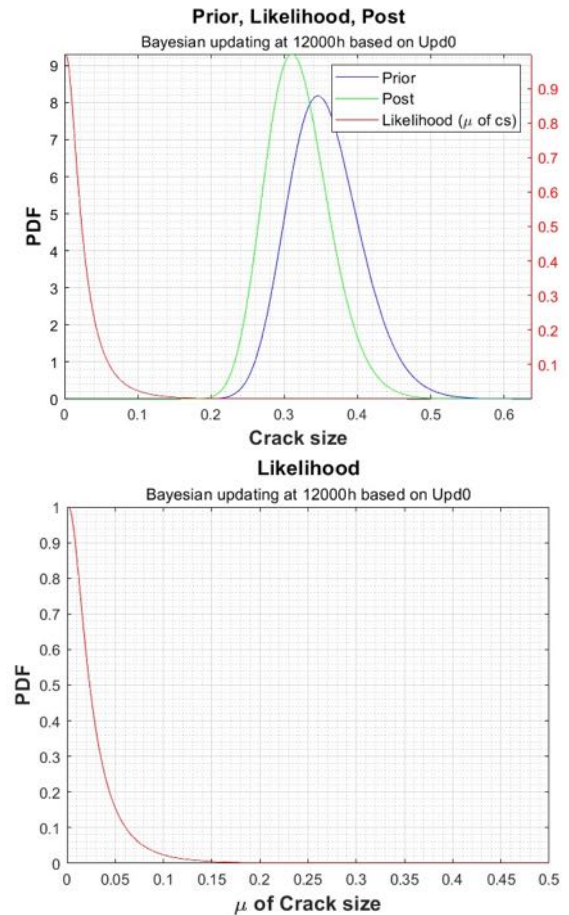
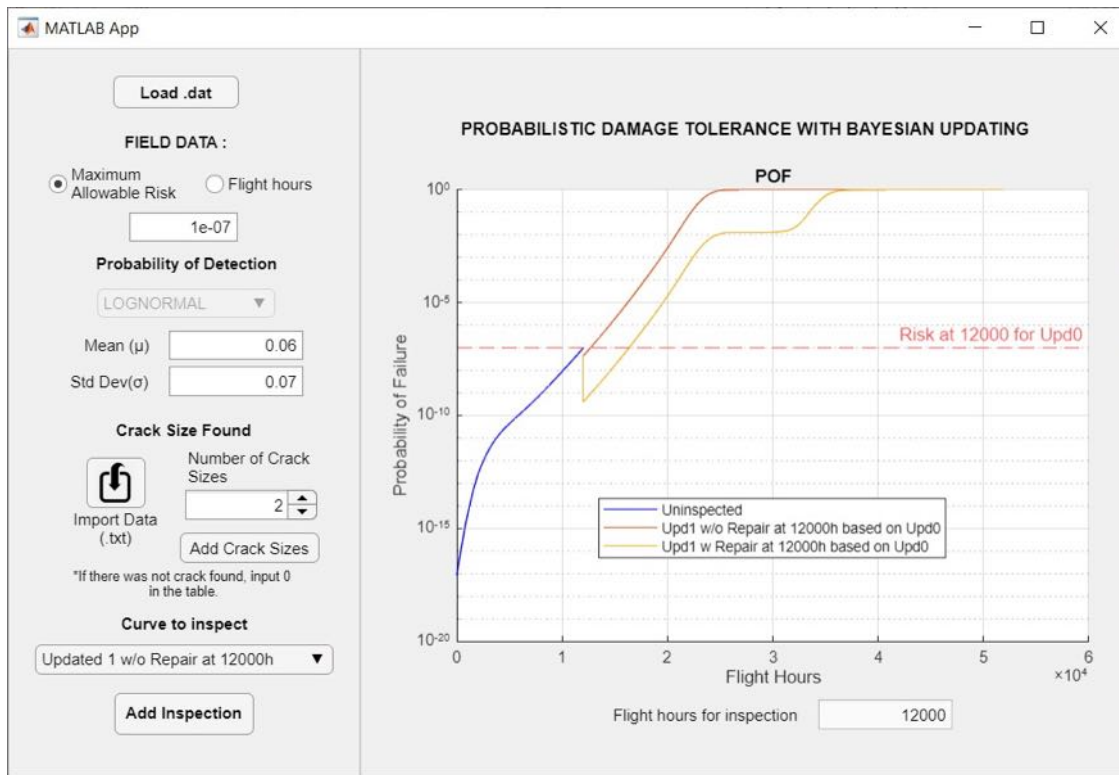


5



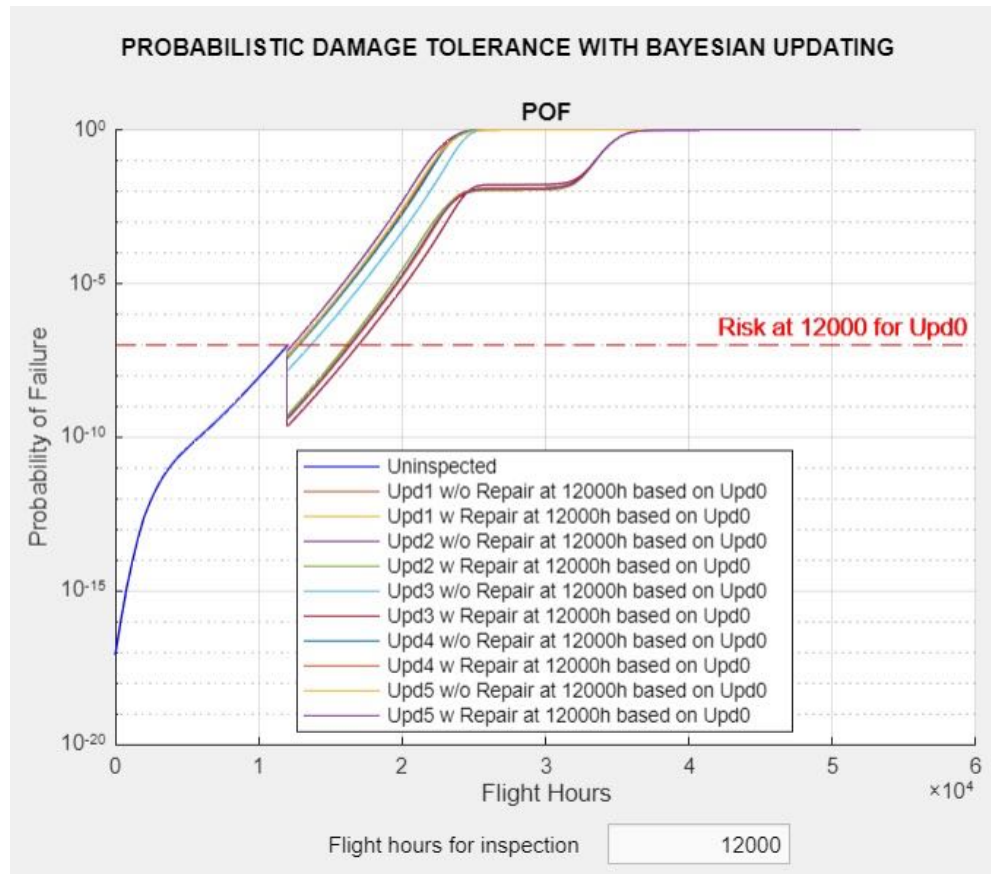


Results Crack size ND-ND





SUMMARY





Conclusions



- A Bayesian updating methodology was integrated within the FAA-sponsored SMART|DT.
 - Bayesian code can also be used as a stand alone code.
- A “finding” or “no finding” can be used to update the PDTA distribution modeling assumptions – Not limited on the number of “finding” or “no finding”.
- Bayesian updating provides a powerful tool to incorporate inspection data into the PDTA/DT analysis.



Acknowledgments



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- Sohrob Mattaghi (FAA Tech Center) – Program Manager
- Michael Reyer (Kansas City) – Sponsor



Thank you



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